USN

Seventh Semester B.E. Degree Examination, Dec.2015/Jan.2016 Aircraft Stability and Control

Time: 3 hrs. Max. Marks: 100

Note: Answer FIVE full questions, selecting at least TWO questions from each part.

PART - A

1 a. Derive an expression for tail contribution $\left(\frac{C_M}{C_L}\right)_{tail}$ for the static longitudinal stability of an

airplane and discuss the downwash at the tail.

(10 Marks)

- b. Define the following terms with equation and graphs:
 - i) Equilibrium condition

iii) Stability criteria

ii) Dynamic stability

iv) Static stability

(10 Marks)

- 2 a. Write the expression for stick-fixed neutral point and discuss the C.G. range on the aircraft.
 - b. Explain longitudinal control and derive the equation for elevator angle versus equilibrium lift coefficient. (10 Marks)
- 3 a. With a help of diagram and expression, explain the control surface floating characteristics and aerodynamic balance. (10 Marks)
 - b. Derive the equation for stick-free neutral point.

(10 Marks)

- 4 a. Briefly explain the effect of wing sweep, flaps and power on dihedral effect. (10 Marks)
 - b. Derive the equation for aileron control force.

(10 Marks)

PART - B

- 5 a. Define directional stability and explain static directional stability with rudder fixed.
 - (10 Marks)

b. Explain rudder lock and one engine in operative condition.

(10 Marks)

(14 Marks)

- **6** a. Derive the equation for longitudinal motion.
 - b. Briefly explain the following with relevant sketch:
 - i) Phugoid mode
 - ii) Short period mode

(06 Marks)

- 7 a. Describe the aerodynamic response to Aileron with adverse yaw effect with required equation and graphs. (10 Marks)
 - b. Derive an expression for change in forward velocity.

- (10 Marks)
- 8 a. Explain Routh's criteria and factors affecting period and damping of oscillation. (10 Marks)
 - b. Explain the following:
 - i) Dutch Roll
 - ii) Spiral instability
 - iii) Auto-rotation and spin stability

(10 Marks)

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